U.S. App. No.: <u>10/780618</u>

IN THE CLAIMS:

Kindly rewrite Claim 1 as follows:

1. (Currently Amended) A method for operating a partially closed, turbocharged gas turbine cycle, the method comprising:

burning fuel in a combustion chamber while supplying a gaseous, compressed working medium which contains combustion air, to generate hot combustion gases;

expanding a working medium which contains the hot combustion gases in a turbine of a gas turbine, said turbine performing work;

extracting heat from the expanded working medium in a downstream recuperator to generate cooled working medium;

compressing the cooled working medium in a compressor of the gas turbine;

feeding heat to the compressed working medium in the recuperator before said compressed working medium re-enters the combustion chamber;

removing a portion of the expanded working medium on a low-pressure side of the recuperator at a removal location which is at a first temperature level, and further expanding said removed expanded working medium portion in the turbine of a first exhaust-gas turbocharger;

sucking in and compressing air with a compressor of the first exhaust-gas turbocharger; and

feeding the compressed air to the working medium on-a the low-pressure side of the recuperator at a feed location which is at a second temperature level;

wherein said gas turbine compressor comprises a radial compressor.